

SCADE User Group Conference, 15.10.2015

Developing Software for the A350 XWB Slat Flap Control Computer with SCADE



Paul Linder, Diehl Aerospace



1 Company Presentation

2 Introduction to the A350 XWB SFCC

3 Development Procedure

4 Modeling Guidelines and Verification Methods

5 Experiences

Corporate Division

DIEHL
Aerosystems

Sales: over € 1,010 m | **Employees:** ≈ 4,700 | **Headquarters:** Laupheim, Germany

Operational Units

DIEHL
Aerospace

Sales: ≈ € 300 m

Employees: ≈ 1,200

Headquarters: Überlingen, Germany

Shareholders: 51% Diehl, 49% Thales

joint venture with **THALES**

DIEHL
Comfort Modules

AOA


DIEHL
Aircabin

DIEHL
Service Modules

Numbers are based on forecast 2015

Flight Control



- Slat Flap Control Computer
- Flaps Lever
- Position Pick-Off Unit

Doors & Slides Management System



- Doors & Slides Management Control Unit
- Local Door Controller
- Autonomous Standby Power Supply Unit
- Control Panels & Indicators
- Sensing
- Swivel Actuator

Integrated Modular Avionics



- Core Processing Input/Output Module (CPIOM)
- Standardized hardware module, I/O capabilities & mechanical packaging
- IMA Tool Suite

Lighting & Interior Functions



- Cabin Lighting Systems
- Cabin Mood Lighting Systems
- Emergency Lighting Systems
- Starlight Systems
- Noise Masking Systems
- Full Automatic Hat Rack Systems

Civil



- A300/310 Family
- A320 Family
- A330/340 Family
- A380 Family
- A350 XWB Family



- 737 Family
- 747 Family
- 767 Family
- 777 Family
- 787 Family

BOMBARDIER

- Bombardier Q400
- Global 7000/8000



- E170/190
- E135/140
- Legacy 600

Military

- A400M
- Eurofighter
- KC-46A Tanker
- NH90
- Tiger
- Tornado





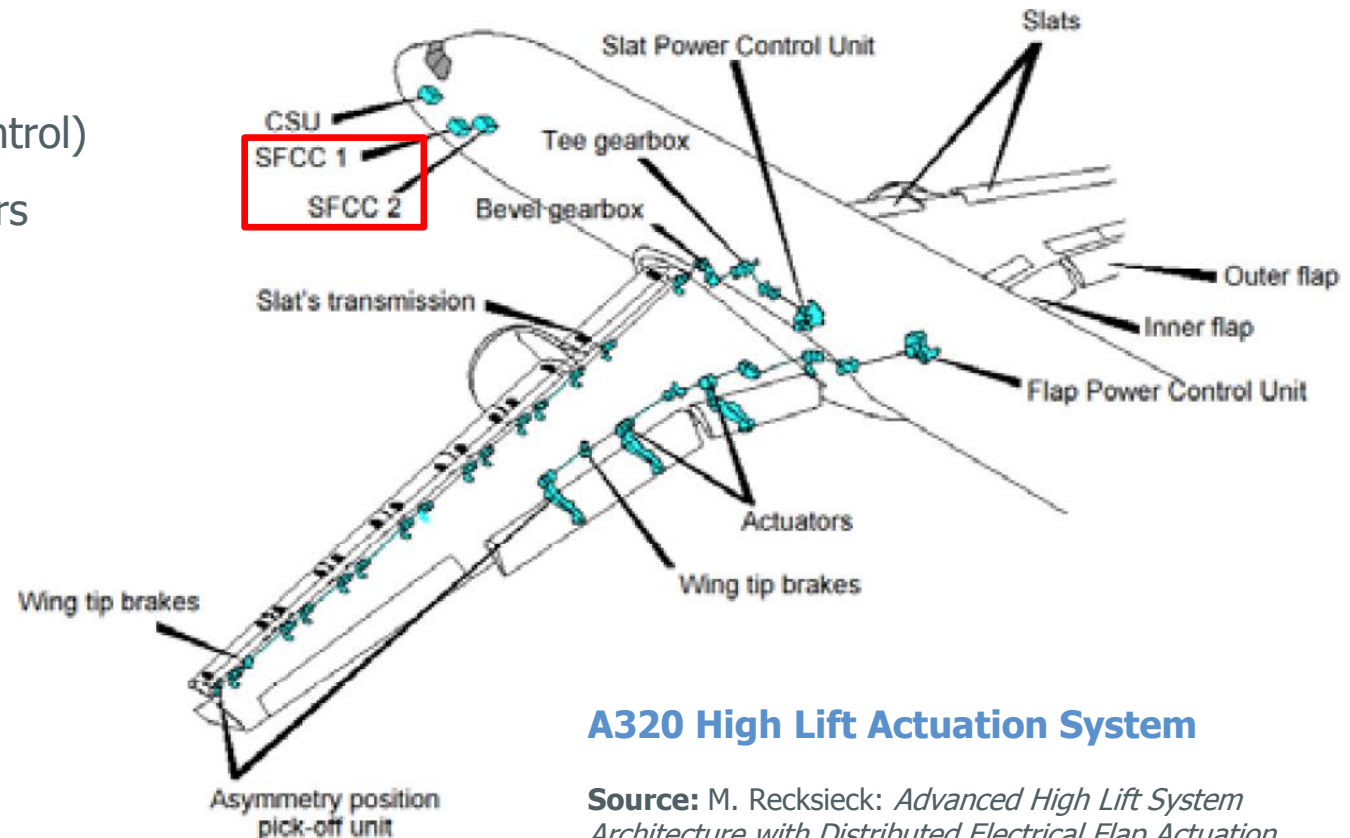
- 1 Company Presentation
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- Slat Flap Control Computer (SFCC)

- Safety-related fly-by-wire system (secondary flight control)
- Controls and monitors high lift system

- High lift system

- Increases lift for take-off and landing

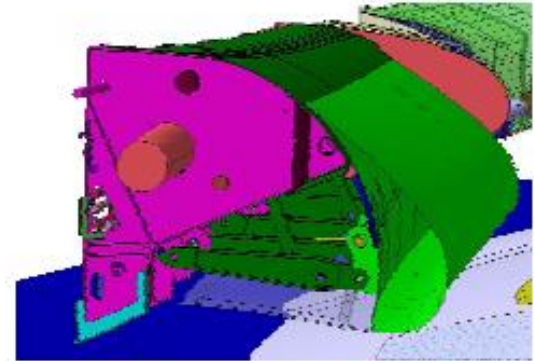


A320 High Lift Actuation System

Source: M. Recksieck: *Advanced High Lift System Architecture with Distributed Electrical Flap Actuation*. Workshop on Aviation System Technology (AST) 2009.

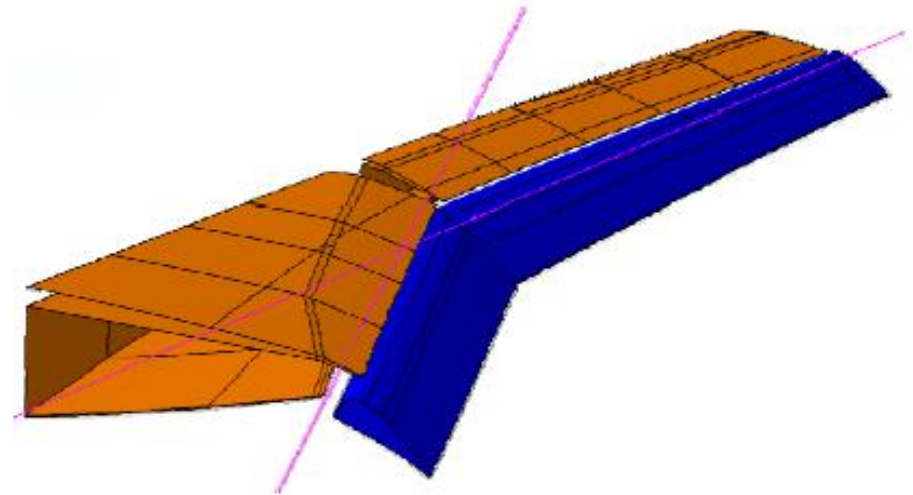
- Technologies

- Droop-nose device on inboard wing
- Multifunctional trailing edge flap system:
Adaptive Dropped Hinge Flap
- Integrated use as high-lift device and for in-flight adaptation of cruise wing shape



- Benefits

- Fuel burn reduction through drag saving
- Load alleviation functions and cruise efficiency enhancement



Source: D. Hills: *The Airbus Challenge* : EADS Engineering Europe, Budapest 9-10th May 08.

- **Functionality**
 - Determination and control of surface position including load alleviation functions
 - Monitoring of high lift system and components (e.g. power control unit)
 - Test functions and maintenance services (BITE)
 - AFDX data loading for SW update
- **Design**
 - 2 exchangeable SFCCs with 2 independent channels (slat/flap) per SFCC
 - Redundant and dissimilar design
 - Overall 16 micro controllers and several DSPs
 - Level A design assurance



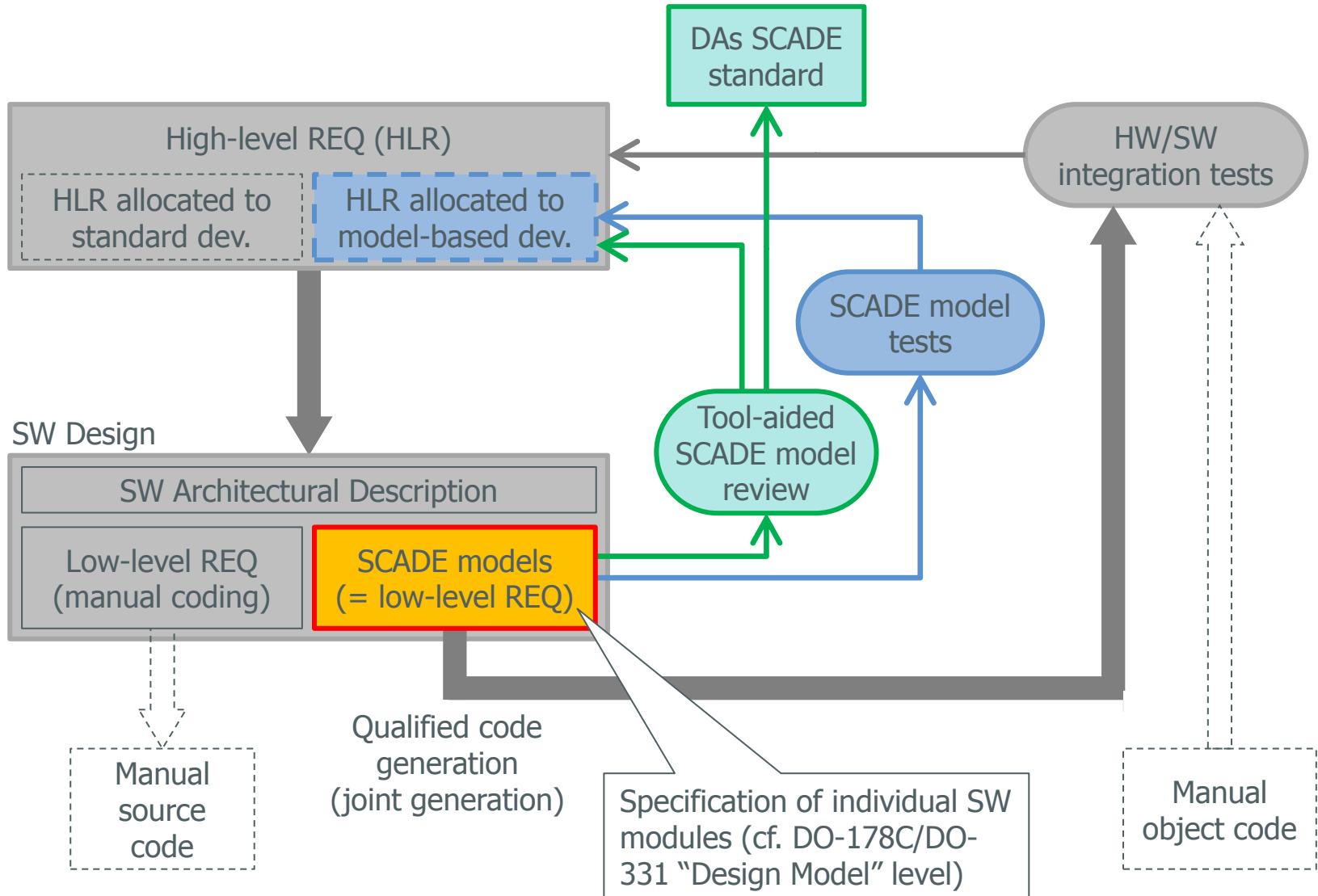
Note: A350 XWB SFCC similar to depicted A380 SFCC.



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- Project context
 - Equipment development project according to ARP-4754 / DO-254 / DO-178B level A
 - Schedule DAs: 07/2008 – ongoing (type certification on 30.09.2014)
- SCADE involvement
 - SCADE applied for level A development of SFCC application SW
 - » Parallel to development of manually coded basic software (e.g. scheduling, driver, data loading)
 - » ~150 application SW modules (e.g. high-lift system monitors, component monitors)
 - SCADE version 5.1 applied
 - » Only data flow diagrams
 - » No state-charts (due to tool qualification constraints), no higher-order functions

DAs SCADE Development Procedure



- High-level REQ

SRD_OPS-REQ-2298

A Slat Cross Lane Output Monitor failure condition shall be detected if the signals indicated in the table below do not match within the associated **Threshold**:

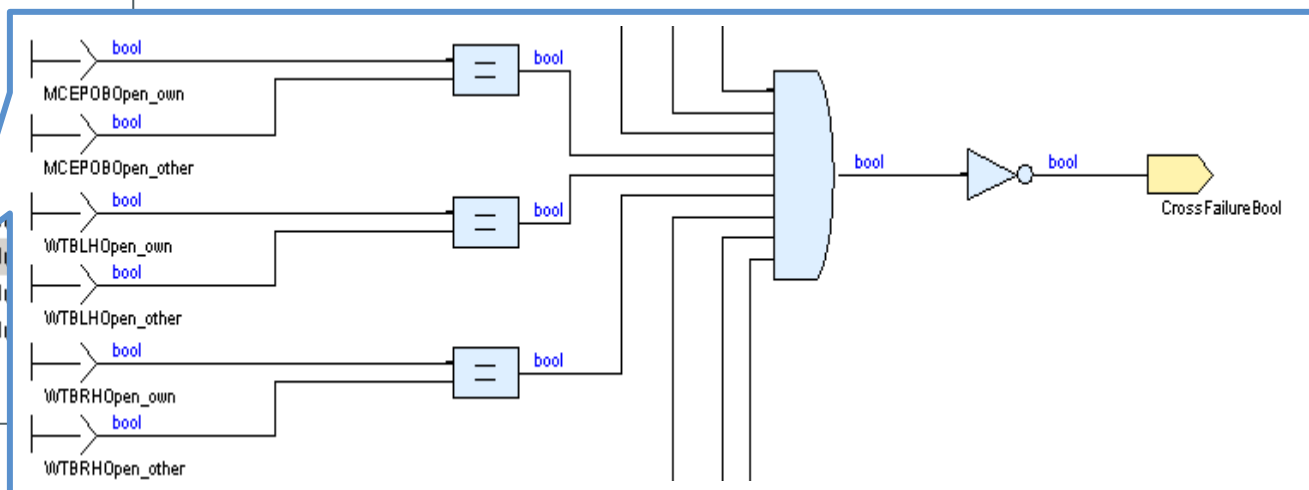
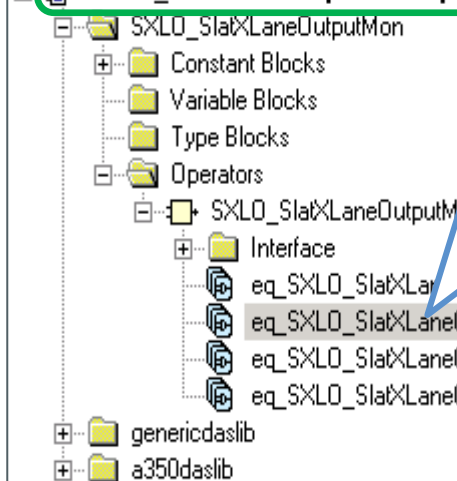
Signal	Description
--------	-------------

- Low-level REQ / SCADE model

SDD_OPS-W-CI-2912 The behaviour of the module is specified by the related SCADE model documented by the SCADE project report **SXLO_SlatXLaneOutputMon.rtf, ClearCase version 22.**

SRD_OPS-REQ-2298 SRD_OPS-REQ-2300;SRD_OPS-REQ-2301.SDD_OPS-REQ-

SXLO_SlatXLaneOutputMon.etp



- High-level REQ

SRD_OPS-REQ-2300

A Cross Lane Output Monitoring failure condition shall be validated if a cross lane failure condition is validated for five (5) validation cycles.

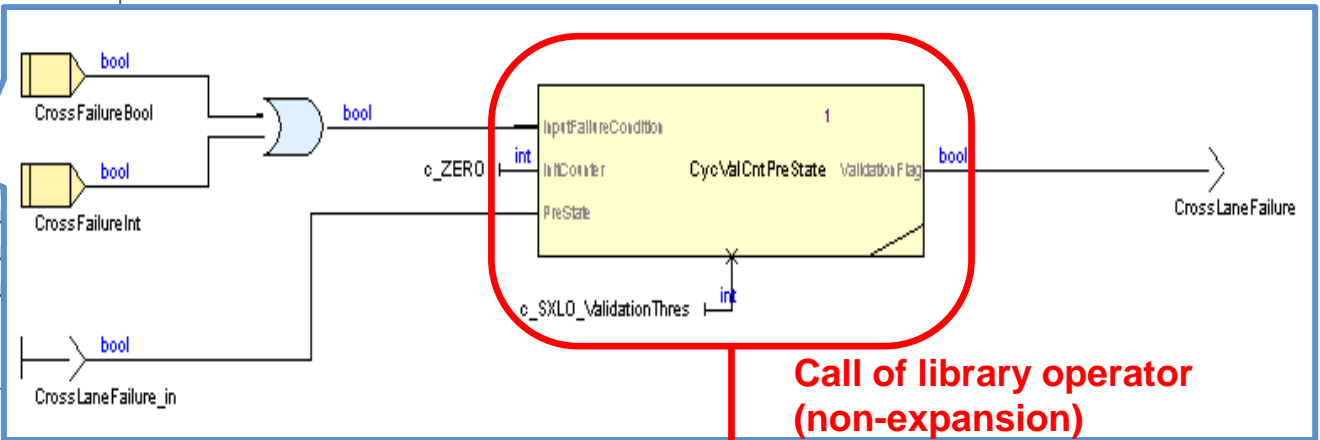
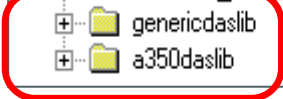
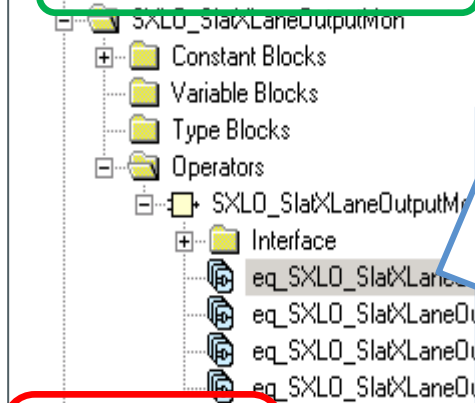
- Low-level REQ / SCADE model

SDD_OPS-W-CI-2912

The behaviour of the module is specified by the related SCADE model documented by the SCADE project report **SXLO_SlatXLaneOutputMon.rtf**, ClearCase version 22.

SRD_OPS-REQ-2298;SRD_OPS-REQ-2300;SRD_OPS-REQ-2301;SRD_OPS-REQ-2302

SXLO_SlatXLaneOutputMon.etp



Call of library operator (non-expansion)

DAs Libraries



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- Guidance on following issues:
 - Tool settings and options to ensure conditions imposed by SCADE tool qualification
 - » E.g. interdiction of unary minus operator to avoid SCADE 5.1 maintenance issue CR ID 5137
 - Modeling conventions to support DAs model verification procedures
 - » E.g. naming and traceability conventions, complexity restrictions, algorithmic constraints
- Overview of rules
 - 16 mandatory rules to avoid undefined and failure-prone features (cf. tool qualification)
 - 23 required rules related to modeling conventions (cf. verification procedures) → Justifications allowed
 - No optional or recommended rules applied

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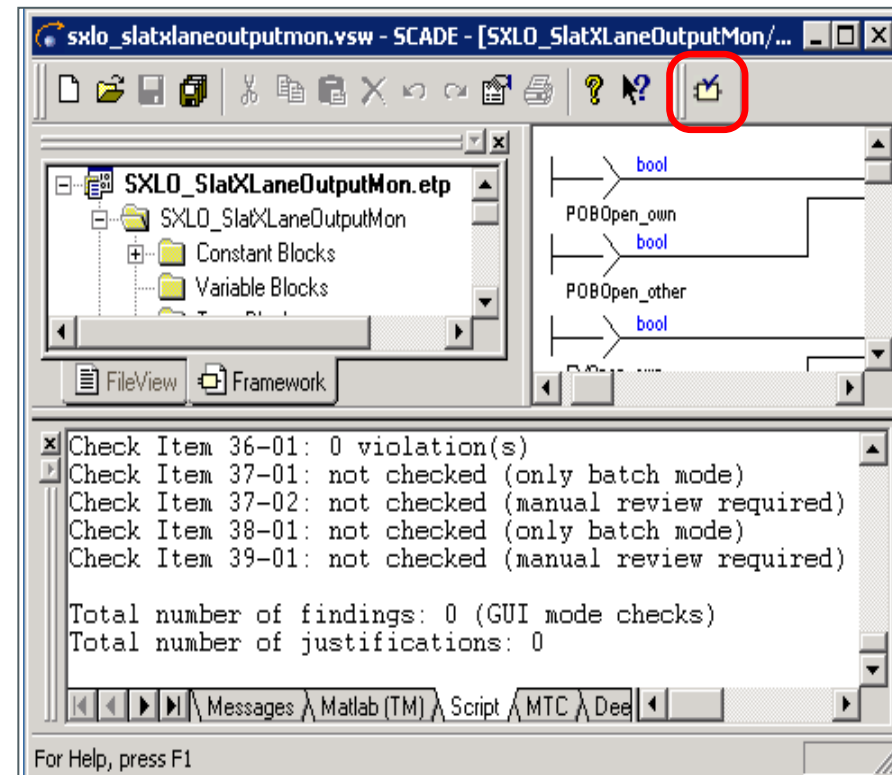
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Seite/Page	1/58
Datei/File Name	PD1000006084_000_07_SCDSTD.docx

SCADE Development Standard (SCDSTD)
for
A350 XWB Slat Flap Control Computer and Sensors (A350 XWB SFCC)
(ATA 27)
for the
Airbus A350XWB Program

	Name/Name Fonkt./Function	Datum/Date	Unterschrift/Signature
Erstellt / Prepared	Paul Linder Software Engineer	15. 8. 2013	Paul Linder
Technisch verifiziert / Technical checked			
Freigelegt / Quality approved			
Technisch Autorisiert / Technical Authorized			

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#00000000_000_07_SCDSTD.docx 6/8/2013 11:00:00

- Automatic check of 26 rules of the DAs SCADE Development Standard
 - Checks generation options, modeling elements, complexity restrictions, naming conventions, model/report/autocode consistency
 - Remaining 13 rules subject to manual review (based on SCADE report)
- Developed with TCL and Python
 - TCL scripts using SCADE API
 - » E.g. **MapRole \$model node**
CountForbiddenModelOperators
 - Python checking source/report generation and producing HTML report
- Qualified as verification tool
 - Qualified “batch mode”
 - Engineering “GUI mode” (see figure)



- HTML report

Checking report

Overview

SCADE StyleChecker, Version 2.0, (c) Diehl Aerospace GmbH 2013

Checking Run Overview

Checking Run: 01 Jun 2015 16:58:25

Summary

Total number of SCADE projects checked: 151
 SCADE projects PASSED: 151
 SCADE projects FAILED: 0
 SCADE projects ERROR: 0

Total number of findings: 38
 Number of mandatory rule findings: 0
 Number of required rule findings: 38
 Number of justifications for required rules: 38

Configuration

Directories searched for SCADE projects:

Overview

SCADE Project	Author	Verdict	Mandatory Findings	Required
fsam_fsasymmonfailval		PASSED	0	1
ssam_ssasymmonfailval		PASSED	0	0
fpal_fadgbhlpwrvailfail		PASSED	0	0
fpar_fadgbrhpwrvailfail		PASSED	0	0
szlo_slatlaneoutputmon		PASSED	0	0

SCADE StyleChecker, Version 2.0, (c) Diehl Aerospace GmbH 2013

Checking Report: Model szlo_slatlaneoutputmon

Checking Run: 01 Jun 2015 16:38:21

Summary

Total number of findings: 0
 Number of mandatory rule findings: 0
 Number of required rule findings: 0
 Number of justifications for required rules: 0

Verdict: PASSED

Configuration

Reference: SCADE Development Standard (SCDSTD) for A350 XWB Slat Flap Control Computer and Sensors (A350 XWB SFCC), Issue 000_07

Checking Objects:

T:\A350_SFCC_SW\models\scade_gen\scade_gen.bat

Checking Protocol

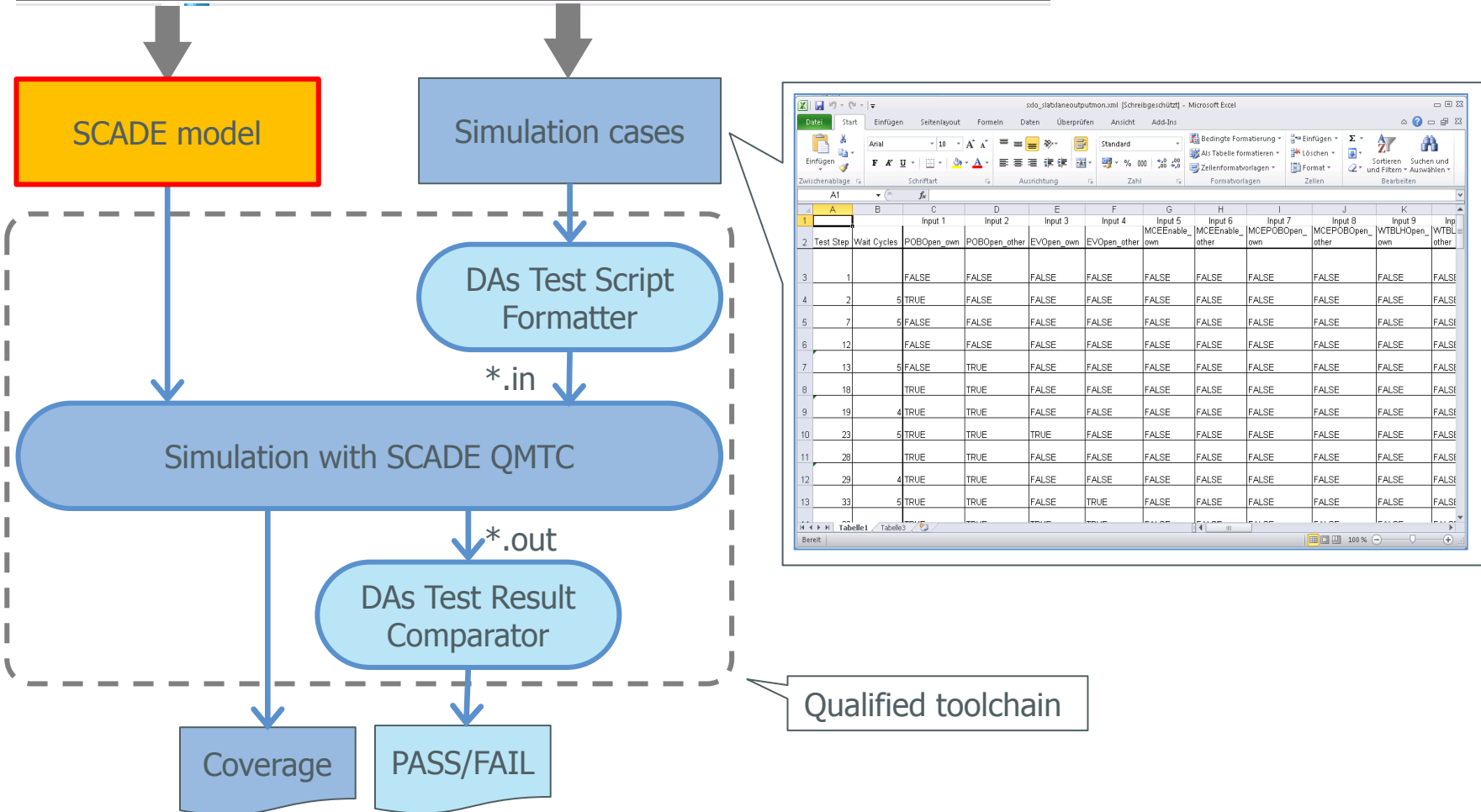
Check Item 01-01: 0 violation(s)
 Check Item 02-01: not checked (manual review required)
 Check Item 02-02: 0 violation(s)
 Check Item 03-01: 0 violation(s)
 Check Item 03-02: 0 violation(s)
 Check Item 04-01: 0 violation(s)
 Check Item 07-01: 0 violation(s)

DAs Model Testing Procedure

SRD_OPS-
REQ-2300

A Cross Lane Output Monitoring failure condition shall be validated if a cross lane failure condition is validated for five (5) validation cycles.

High-level requirement








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- Successful certification of level A software!
 - EASA type certification Airbus A350 XWB on 30.09.2014
- Estimated >2x higher efficiency for SW module development
 - Omission of source code verification due to qualified source code generation
 - Bypass of effort-consuming conventional LLR specification and module testing
- Automatic consistency checks proved very valuable



- Set model expansion options in conformance to testing approach 
 - 100% structural coverage may not be achieved with full expansion of libraries
 - Advice: Non-trivial library operators should not be expanded
- Mind the configuration management 
 - Not only SCADE model and higher level requirements but also traceability data and review results (findings) have to be subject to version control
- Be aware of your modeling semantics 
 - Identical syntax may have different meaning on different specification levels (cf. DO-178C/DO-331 “Design Model” vs. “Specification Model”)
 - Do not disregard quality conditions and design constraints requirements

Contact

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